



EU AERONAUTICS RESEARCH CO-OPERATION



SE Ivchenko-Progress



INTERNATIONAL RECOGNITION OF CERTIFICATION AUTHORITIES





Quality Systems Certification Authority of "QUALITY" Certification Centre (Russian Federation) Certificate of Compliance POCC UA.MX04.B00034 POCC UA.MX04.B00043 POCC UA.MX04.B00044



Bureau Veritas (France) Certificate No. 213617, No. 010-UKR and No. 010-UKF



Aviation Register of Interstate Aviation Committee (ARMAK) Certificates No. SPR-15, No. R-3, No. R-69 and others



State Department of Aviation Transport of Ukraine

Certificates No. VR 0036, No. TD 0005 and others

* Totally 59 certificates of various types.

THE BASIC SPHERES OF ACTIVITIES



DESIGN



MANUFACTURE



OVERHAUL



TEST AND DEVELOPMENT



PUTTING IN SERIES PRODUCTION
AND IMPROVEMENT OF
CONSUMER'S CHARACTERISTICS



More than 60 certificates of Bureau Veritas, EASA, Central Civil Aviation Administration of China, IAC AR and GosAviaSluzhba of Ukraine confirm a type design conformity, quality, reliability and the right to design, manufacture, overhaul and upgrade the engines.

DIRECTIONS OF ACTIVITY



An-12



Al-20K Series 5, Al-20M Series 6

An-10



AI-20A, AI-20K Series 5

IL-18



Al-20A, Al-20K Series 5, Al-20M Series 6

Ми-1



АИ-26В

Be-12



AI-20D Series 4

An-24



Al-24 Series 2, Al-24T

Yak-40



Al-25 Series 2

Yak-42D



D-36 Series 1

An-72



D-36 Series 1A

CORPORATION ACTIVITY



IVCHENKO-PROGRESS CORPORATION WAS FOUNDED ON 12.01.2007 THE FOUNDATORS ARE SE "IVCHENKO-PROGRESS" and JSC "MOTOR SICH"



















Milestones in development of SE Ivchenko-Progress gas turbine aero engines



65 years

The aero engines designed by SE Ivchenko-Progress power 54 types of A/C in 122 countries of the world.

Total operating time of GTEs exceeds 300 million hours

4-th stage

Turbofans with high power and thrust: D-136, D-18T

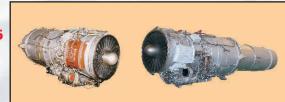
D-27 propfan,
TV3-117VMA-SBM1 turboprop, D-436 turbofan,
AI-22 turbofan, AI-222 turbofan, AI-450 turboshaft,
AI-450 turboprop, SPM-21 turbofan



2-nd stage

Turbofans: AI-25, AI-25ТЛ, D-36 APUs: AI-9, AI-9V

3-rd stage



1-st stage

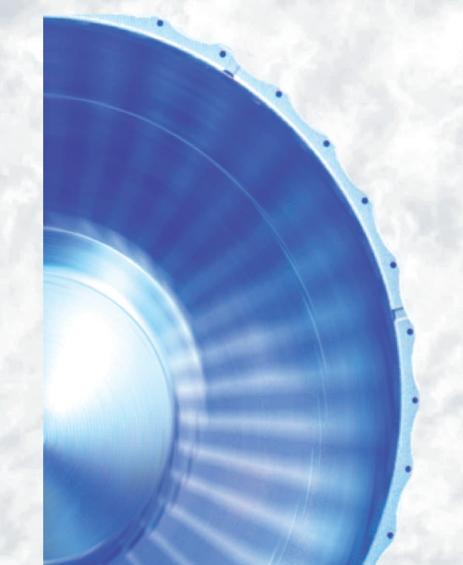
Turboprops: AI-20, AI-24 APU: AI-8







Piston engines: AI-26, AI-14, AI-4





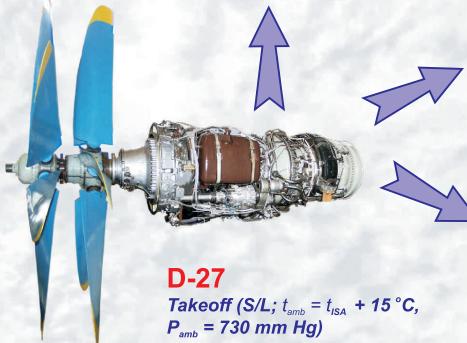
Advanced engines

D-27









 N_e , ehp 14 000 C_N , kg/h/ehp 0.170

An-70,-70T

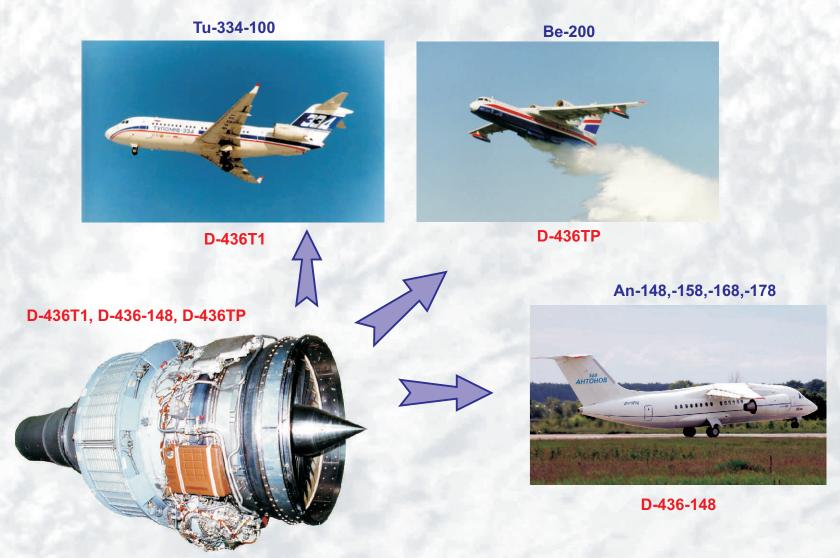


A-42PE



D-436T1, D-436-148, D-436TP





Takeoff (S/L static; ISA)

R, кgf C_R, кг/kgf·h **D-436T1**7 650
0,364

7 650 0,356

7 000 0,351

D-436TP D-436-148

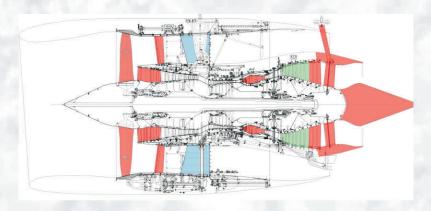
7500

D-436-148FM

D-18T series 3M, D-18T series 5



D-18T series 3M

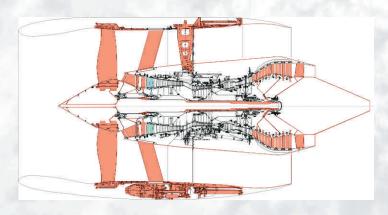


R, kgf 23430





D-18T series 5





R, kgf 27500

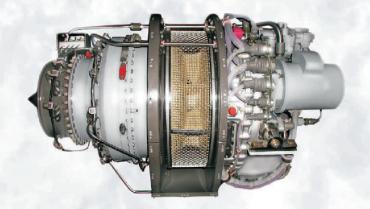
An-124-300



AI-450, AI-450M, AI-450-2



AI-450



Helicopters with takeoff weight of 3100...3400 kg



Takeoff (S/L static; ISA)
Al-450

N, hp 465

AI-450M



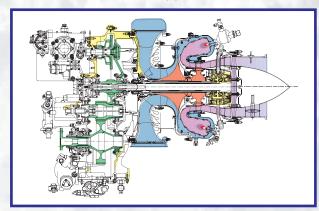
Mi-2M



Takeoff (S/L static; ISA)
Al-450M

N, hp 400 $C_N, kg/h/hp$ 0.260

AI-450-2



Helicopters with takeoff weight of 3000...3300 kg



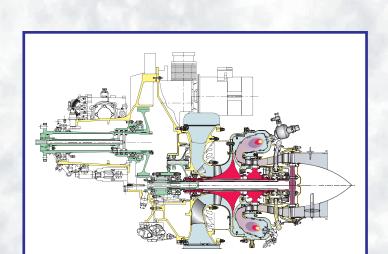
Takeoff (S/L static; ISA)

AI-450-2

N, hp 630...730 C_N,kg/h/hp 0.255

AI-450S





N, hp

Yak-18T



UAV





Light multi-purpose aircraft



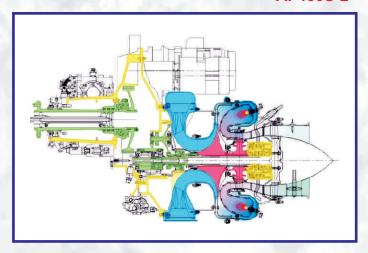
AI-450S

Takeoff (S/L static; ISA) 400...550

AI-450S-2, AI-450BP, -450BP-2



AI-450S-2



EV-55



Light multi-purpose aircraft

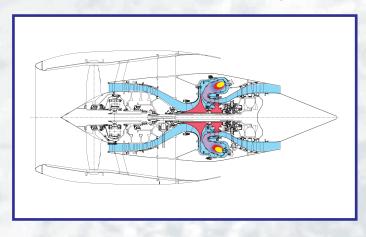


AI-450S-2

Takeoff (S/L static; ISA) N, hp

630...730

AI-450BP, -450BP-2



UAV



Takeoff (S/L static; ISA)

AI-450BP, -450BP-2

R, kgf 410...560 C_R, kg/h/kgf 0.370





PROPULSION INTEGRATION

ADVANCED STRUCTURE OF SMALL GAS TURBINE ENGINE

THE PROJECT CONSORTIUM 39 participants,14 countries

Effectiveness of the small aircraft, its competitiveness, affordable operation costs and safety directly depends on the efficiency of the small turboprop engine which powered the aircraft

- Decreasing of weight of the power unit for 6 8%
- Decreasing of fuel consumption for 7-12 %
- Decreasing of overall dimensions of the power unit
- Extension of engine service life and its systems for 10 15 %
- Decreasing of maintenance costs for 7 9 %





Small engine test bench Small engine rotor design optimisation facilities`for prospective design experimental investigations



Total budget: 33,7 MIL. EUR EC contribution: 18,1 MIL. EUR



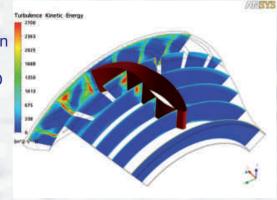


design optimisation cooling system

design

Small turbine blade Small turbine blade Small turbine blade external flow and cooling system CFD calculations

Design optimization of interturbine diffuser by CFD calculations





Thank you for your attention

